



Determination of the 1 in 10, 1 in 50 and 1 in 100 Year Space Weather Event

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Motivation

- Modern satellites have a life expectancy of 10-20 years.
- Satellite operators and engineers therefore require realistic estimates of the worst case environments that may occur on these and longer timescales.
- Satellite insurers also require this information to help them evaluate realistic disaster scenarios.





Extreme Space Weather Events

- As part of SPACESTORM we have conducted extreme value analyses to determine the 1 in 10, 1 in 50, and 1 in 100 year space weather event for
 - relativistic electrons at GEO and HEO
 - internal charging currents at MEO
 - energetic electrons at LEO







Extreme Value Analysis

- Two main methods for extreme value analysis
 - block maxima
 - exceedances over a high threshold
- The exceedances over the threshold approach makes the best use of the available data and has been successfully applied in many fields
- For this approach the appropriate distribution function is the Generalised Pareto Distribution (GPD)





Declustering

- Values can exceed the threshold on consecutive measurements
- The statistical analysis requires that the individual exceedances are independent
- Technique to deal with this is known as declustering





Declustering

- Use an empirical rule to define clusters of exceedances depending on the temporal behaviour of the data
- Identify the maximum excess in each cluster
- Fit the GPD to the cluster maxima





Generalised Pareto Distribution

• The GPD may be written in the form

$$G(x-u) = 1 - (1 + \xi(x-u)/\sigma)^{-1/\xi}$$

where: x are the cluster maxima above the chosen threshold u ξ is the shape parameter which controls the behaviour of the tail

 σ is the scale parameter which determines the dispersion or spread of the distribution

• We fit the GPD to the tail of the distribution using maximum likelihood estimation





Determination of the 1 in N Year Event

- Our major objective is to determine the 1 in N year space weather event
- The value that is exceeded on average once every N years can be expressed in terms of the fitted parameters σ and ξ as:

$$x_{N} = u + (\sigma/\xi)(Nn_{d}n_{c}/n_{tot})^{\xi} - 1))$$

where n_d is the number of data points in a given year, n_c is the number of cluster maxima and n_{tot} is the total number of data points





Relativistic Electrons at GEO

- For this study we use data from the EPS sensors on board the NOAA GOES satellites at GEO
- We use E > 2 MeV electron data from 1st January 1995 to 30th June 2014
- Study uses data from GOES 8, 9, 10, 11, 12, 13 and 15



credit: NOAA

Typical Orbital Parameters Altitude: 35,800 km Inclination: 0°





Relativistic Electrons at GEO

- Electron data
 - have been corrected for proton contamination
 - for the first time the data have been corrected for dead time
 - dead time correction ranges from a factor of 1.0-1.15 for fluxes around 5000 cm⁻²s⁻¹sr⁻¹ to ~2 for the largest fluxes observed



credit: NOAA

Typical Orbital Parameters Altitude: 35,800 km Inclination: 0°





GOES West: Return Level Plot

- 1 in 10 Year Flux
 - 1.84x10⁵ cm⁻²s⁻¹sr⁻¹







GOES West: Return Level Plot

- 1 in 10 Year Flux
 - 1.84x10⁵ cm⁻²s⁻¹sr⁻¹
- 1 in 100 Year Flux
 - 7.68x10⁵ cm⁻²s⁻¹sr⁻¹







GOES West: Return Level Plot

 Largest observed flux is a 1 in 50 year event







Comparison with Koons [2001] Study

- Our results are significantly larger than those presented in Koons [2001]
- The 1 in 10 year event at GOES West is about a factor of 2.7 times that estimated by Koons [2001]
- For more extreme events, the 1 in 100 year event at GOES West is about a factor of 7 times that estimated by Koons [2001]







Comparison with Koons [2001] Study

- Reasons for differences between studies include
 - use of modern instrumentation
 - correction of data for dead time
 - declustering of data
 - sorting data according to satellite location







Impact

- The revised extreme flux levels have been used to update the UK
 National Risk Assessment
- The results have also been used by a satellite operator in the evaluation of satellite tenders





Extended Analysis (1995-2017)

- Analysis recently updated to include an additional 2.5 years of data
- Data now cover 2 solar cycles







Extended Analysis (1995-2017)

- 1 in 10 Year Flux
 - 1.93x10⁵ cm⁻²s⁻¹sr⁻¹
 - 5% increase
- 1 in 100 Year Flux
 - 5.54x10⁵ cm⁻²s⁻¹sr⁻¹
 - 30% decrease
 - 5 times the Koons [2001] estimate
- Largest event seems more extreme – 1 in 80 year event







Energetic Electrons at LEO

- We use the 2 s resolution E > 30 keV, E > 100 keV, and E > 300 keV MEPED electron data from NOAA15 to NOAA19 from 1 July 1998 to 30 June 2014
- We calculated the maximum flux in each 3 h window as a function of energy and L^{*}



NOAA-19

credit: NOAA

Typical Orbital ParametersAltitude:854 kmInclination:98.7°Period:102.1 min





E > 30 keV Electrons: 1 in N Year Event Levels

- The 1 in 10 year flux of E > 30 keV electrons (black line) shows a gradual increasing trend with L* ranging from 1.8x10⁷ cm⁻²s⁻¹sr⁻¹ at L* = 3.0 to 6.6x10⁷ cm⁻²s⁻¹sr⁻¹ at L* = 8.0
- The 1 in 100 year flux (red line) is generally a factor of 1.1 to 1.5 larger than the corresponding 1 in 10 year event





E > 100 keV Electrons: 1 in N Year Event Levels

- The 1 in 10 year flux of E > 100 keV electrons (black line) peaks at 1.9x10⁷ cm⁻²s⁻¹sr⁻¹ at L* = 4.5–5.0 decreasing to minima of 7.1x10⁶ and 8.7x10⁶ cm⁻²s⁻¹sr⁻¹ at L*= 3.0 and L* = 8.0 respectively
- The 1 in 100 year event is a factor of 1.1 to 3.1 larger than the corresponding 1 in 10 year event





E > 300 keV Electrons: 1 in N Year Event Levels

- In contrast to the E > 30 keV electrons, the 1 in 10 year flux of E > 300 keV electrons shows a general decreasing trend with L^{*} ranging from 2.4x10⁶ cm⁻²s⁻¹sr⁻¹ at L^{*}= 3.0 to 1.2x10⁵ cm⁻²s⁻¹sr⁻¹ at L^{*}= 8.0
- The 1 in 100 year event (red line) is a factor of 1.7 to 5.9 larger than the corresponding 1 in 10 year event





Internal Charging Currents at MEO

- For this study we use data from the SURF internal charging monitor on board ESA's Giove-A spacecraft in MEO
- Use data from 29th December 2005 to 5th January 2016



credit: ESA

Orbital Parameters Altitude: 23,300 km Inclination: 56° Period: 14 hours





SURF Internal Charging Monitor

- SURF is designed to measure the small currents which penetrate spacecraft surfaces and cause internal charging
 - consists of three aluminium collector plates mounted in a stack
 - top, middle and bottom plate respond to electrons with energies greater than 500, 700 and 900 keV respectively



Plate	Threshold	Peak Response
Тор	500 keV	700-900 keV
Middle	700 keV	1.1-1.4 MeV
Bottom	900 keV	1.6-2.0 MeV





Top Plate: 1 in N Year Event Levels

- 1 in 10 year top plate current
 - decrease with L*
 - ranges from 1.0 pAcm⁻² at L*= 4.75 to 0.03 pAcm⁻² at L* = 7.0
- 1 in 100 year top plate current is generally a factor of 1.2 – 1.8 times larger than the 1 in 10 year event







Bottom Plate: 1 in N Year Event Levels

- 1 in 10 year bottom plate current
 - decrease with L*
 - ranges from 0.4 pAcm⁻² at L*= 4.75 to 0.01 pAcm⁻² at L* = 7.0
- 1 in 100 year bottom plate current is generally a factor of 1.4 – 2.6 times larger than the 1 in 10 year event







Comparison with Engineering Design Standards

- Both NASA and the European Cooperation for Space Standardization (ECSS) have guidelines on charging current
 - a maximum average current of 0.1 pAcm⁻² over a 24 hour period is commonly adopted
- For dielectrics operating at temperatures less than 25°C the ECSS have adopted a threshold of 0.02 pAcm⁻²
- For comparison with engineering design standards we repeated the analysis using daily-averaged plate currents over the entire orbit path





1 in N Year Events Averaged Along Orbit Path

- The 1 in 10 year top plate current is a factor of 2.1 times the upper design threshold
- The 1 in 10 year middle and bottom plate currents are equal to the upper design threshold

Plate	1 in 10 year current (pAcm ⁻²)	1 in 100 year current (pAcm ⁻²)
Тор	0.21	0.24
Middle	0.1	0.14
Bottom	0.1	0.16





Updated Analysis

- We recently updated the analysis to include an extra year of data
- We found only small differences in return levels when the extra year was included – of the order 2 – 5 %







Relativistic Electrons from HEO

- We use data from SREM on board ESA's INTEGRAL spacecraft in HEO
- Use data from October 2002 to 31st December 2016





credit: ESA

Orbital Parameters Apogee: 153,000 km Perigee: 10,000 km Inclination: 51.6° Period: 72 h





Relativistic Electrons from HEO

- Flux intensities derived using SREM dedicated inverse scheme developed by Sandberg *et al.* [2012]
- Use data within 15° of magnetic equator
- Determine the 1 in N year space weather events as a function of energy and L*





Orbital Parameters Apogee: 153,000 km Perigee: 10,000 km Inclination: 51.6° Period: 72 h



Integral

$L^* = 4.5$

- 1 in 10 year flux decreases from 1.4x10⁷ cm⁻²s⁻¹sr⁻¹MeV⁻¹ at E = 0.69 MeV to 5.3x10⁵ cm⁻²s⁻¹sr⁻¹MeV⁻¹ at E = 2.05 MeV
- 1 in 100 year flux is a factor of 1.1 to 1.2 times larger than the 1 in 10 year flux







- 1 in 10 year flux decreases from 4.4x10⁶ cm⁻²s⁻¹sr⁻¹MeV⁻¹ at E = 0.69 MeV to 1.2x10⁵ cm⁻²s⁻¹sr⁻¹MeV⁻¹ at E = 2.05 MeV
- 1 in 100 year flux is a factor of 1.1 to 1.4 times larger than the 1 in 10 year flux
- 1 in N year fluxes at GEO a factor of 3-4 less than at equatorial MEO







1 in 10 Year Fluxes as a function of L* and Energy

- Results can be used to determine the 1 in 10 year electron flux as a function of energy and L*
- Results cover the range
 - 0.69 2.05 MeV
 - 4.0 < L* < 6.75







Summary

- The 1 in N year fluxes and plate currents computed as part of this work package serve as benchmarks against which to compare other space weather events in GEO, MEO and LEO
- They can be used to assess how hostile the space weather environment might become in a worst-case scenario.
- The results may also be used to compute the return period of any given space weather event to determine if the event was particularly extreme for any given location.











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