

'Real-time' Modelling of Electrostatic Fields within Dielectrics to Provide a Space Weather Risk Index for Internal Charging

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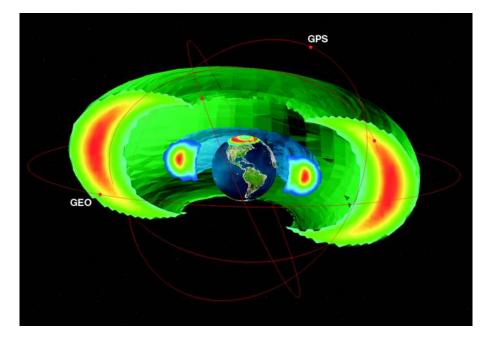
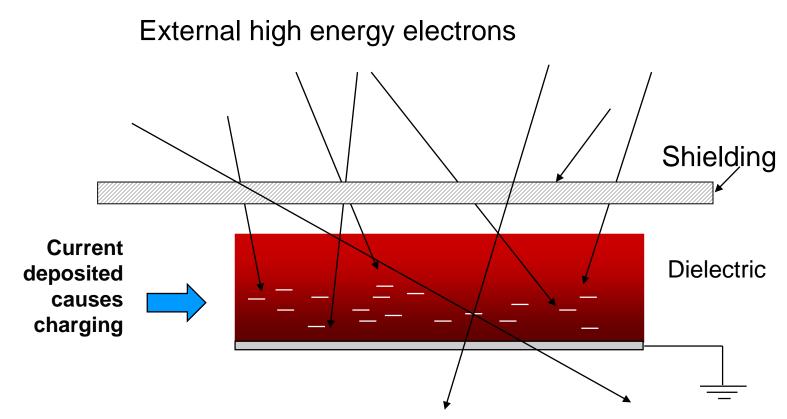


Image: NASA





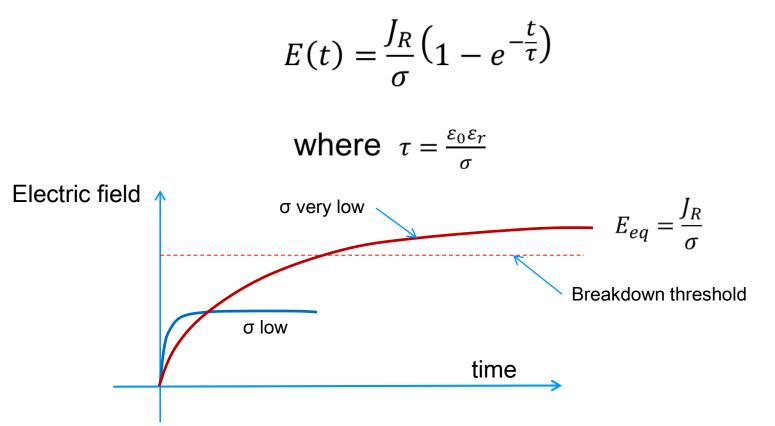
Internal charging



- Internal electric field is the parameter which determines ESD events
- Dielectric breakdown typically at ~10⁷ V/m
- Ideally would have observations of electric fields developed inside the dielectrics
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Internal charging curve

At a given depth and assuming non-varying conductivity:



If σ is small (leading to large equilibrium E-fields) then time constant is long

Modelling electric field in dielectrics

Mike Bodeau* examined charge/field build-up using a 'leaky capacitor' model

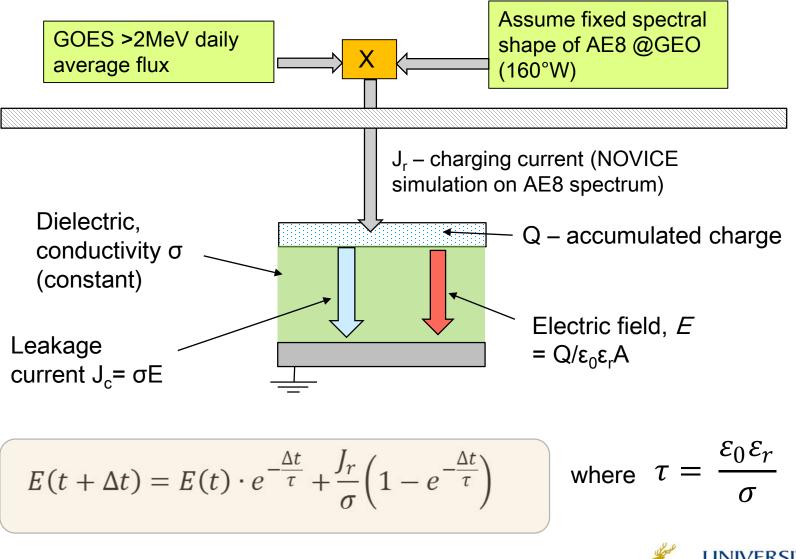
Used GOES record of energetic electron fluxes in GEO to create a 20 year history of electric field/charge density

For long time constant materials the 0.1pA cm²' charging current rule could be insufficient. Maybe be factor of ten lower for some materials.

> *M Bodeau, High Energy Electron Climatology that Supports Deep Charging Risk Assessment in GEO, Proc. of 48th AIAA Aerospace Sciences Mtg, January 2010, Orlando, Florida.



1-D leaky capacitor model



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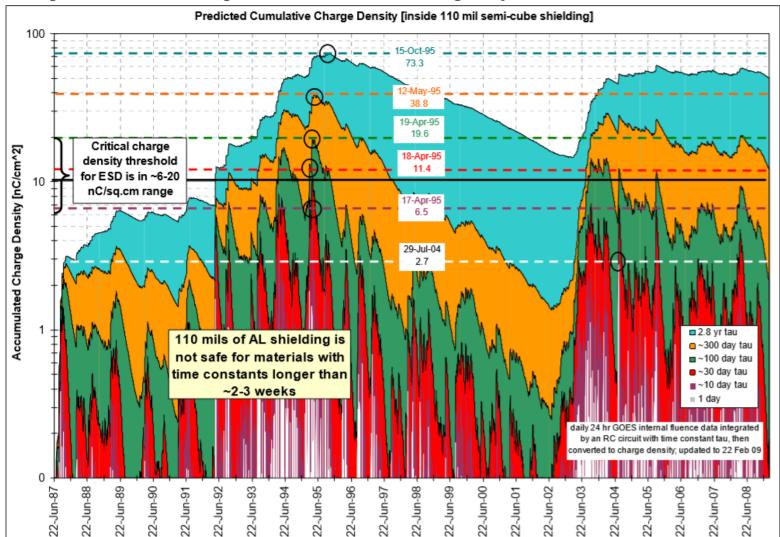


Figure 12. 110-mil Shielding is Unsafe for Materials with Long Decay Time Constant

*M Bodeau, High Energy Electron Climatology that Supports Deep Charging Risk Assessment in GEO, Proc. of 48th AIAA Aerospace Sciences Mtg, January 2010, Orlando, Florida.



Drawbacks of the simple model

Fixed electron spectrum scaled by >2MeV flux

 AE8 is a static long term average spectrum whereas actual one is continually changing

Fixed bulk conductivity

- Radiation induced conductivity in dielectrics not accounted for
- Can time constants really be as long as weeks, months or years?



New developments in this paper

Use in-flight measured deposited currents

- Instead of 2MeV flux + fixed spectrum + radiation transport calculation
- Directly acquired by the SURF instrument

Include radiation induced conductivity (RIC)

• important in many polymers

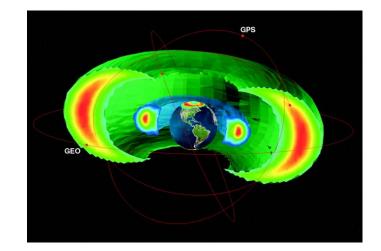
MEO orbit E-field modelling

- critical infrastructure located here (navigation satellites)
- significant internal charging threat
- this is where our measurements are from

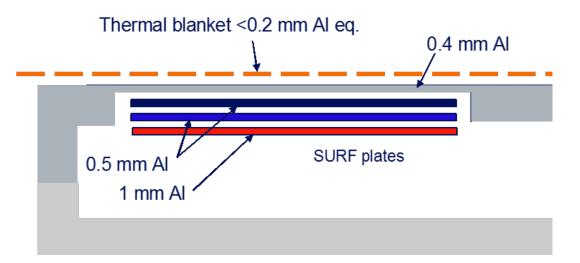


Internal charging current measurements

- December 2005 launch
- Orbit 23,260 km and 56 degrees inclination, 27 month lifetime
- Re-orbited by +300km in 2009
- Still collecting data.

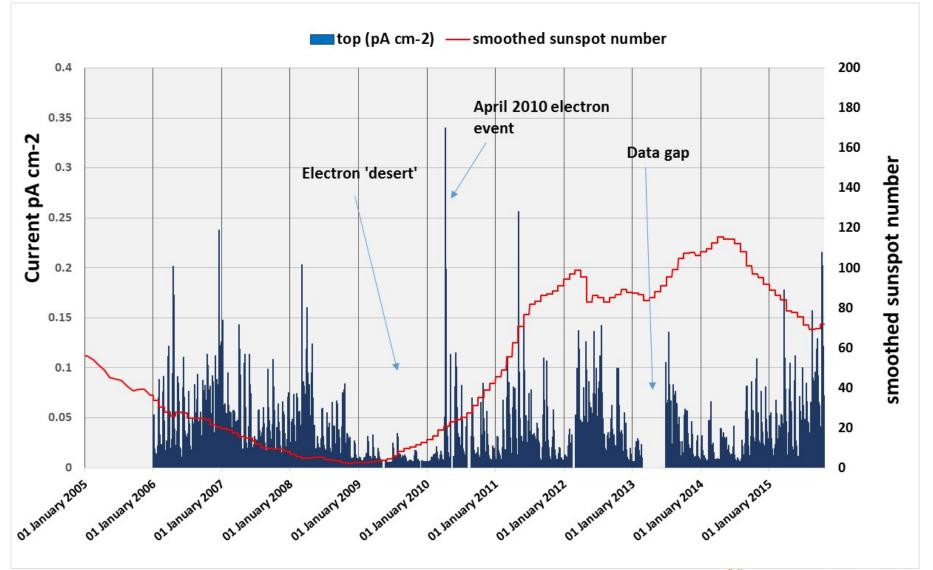






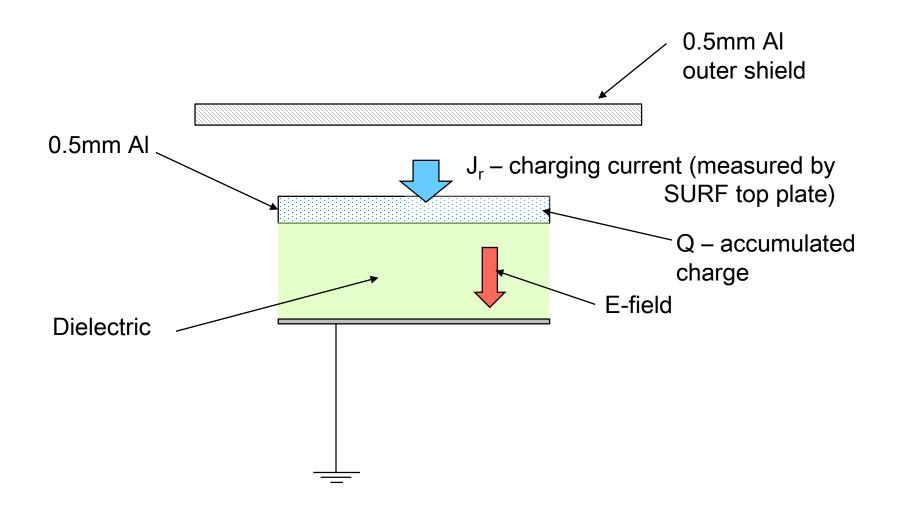


Top plate charging current (daily averages)



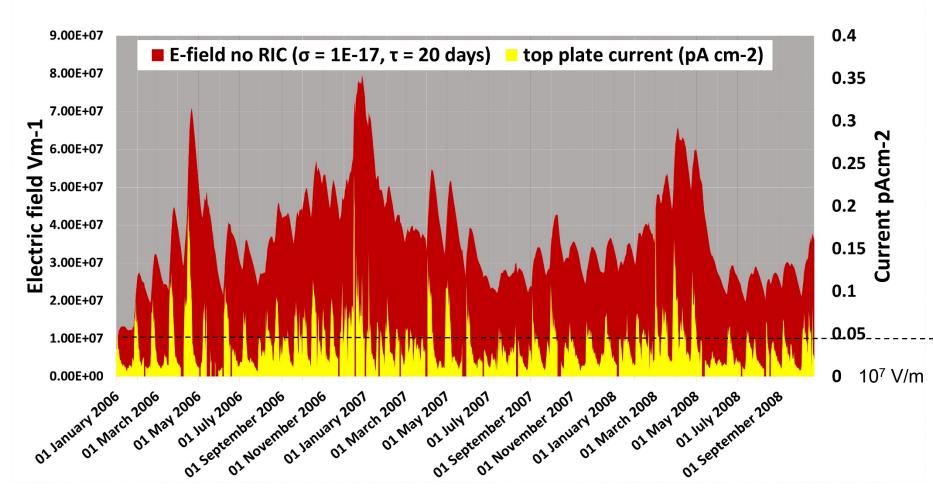


Leaky capacitor model





Dielectric internal field modelling (2006 - 2008)



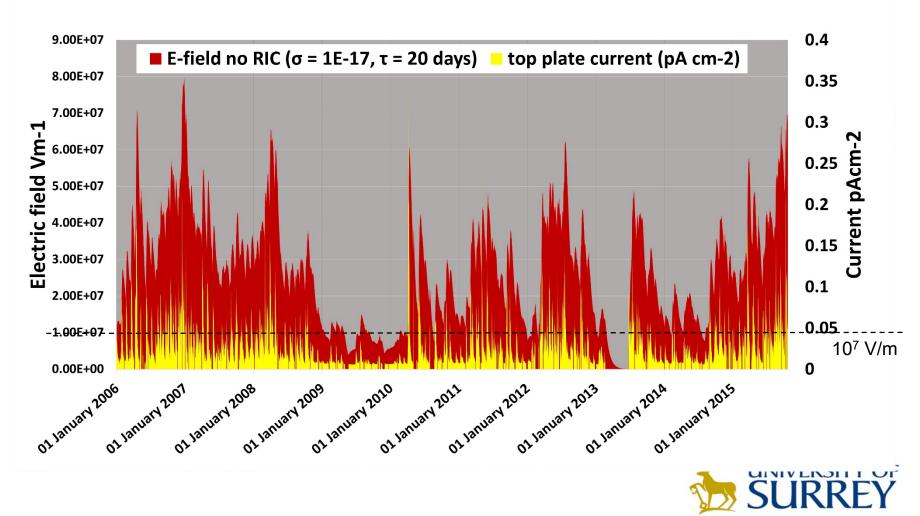
0.5 mm Al shielding; 0.5mm Al-eq. thickness dielectric



Dielectric internal field modelling (2006 - 2009)

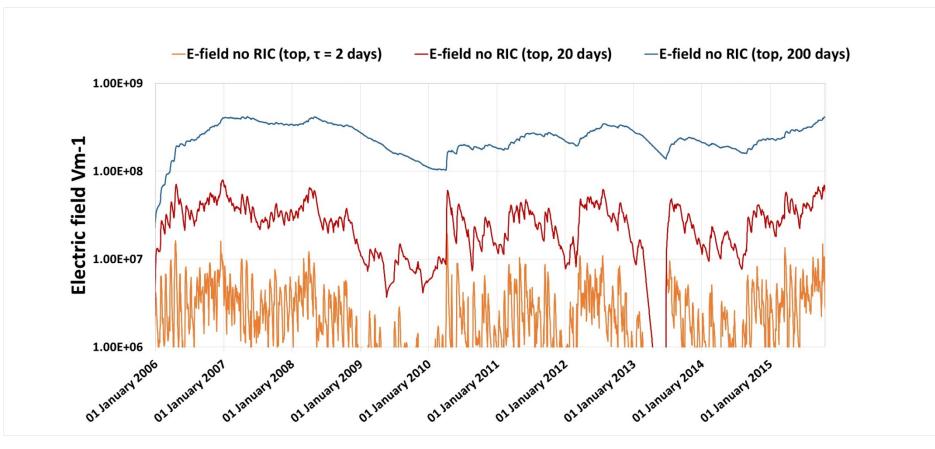
0.5 mm Al shielding

0.5mm Al-eq. thickness absorber



Electric fields (MEO, 0.5mm Al shield, 0.5mm Al-eq absorber, no RIC)

0.5 mm AI shielding; 0.5mm AI-eq. thickness absorber





Radiation induced conductivity (RIC)

Ionising dose leads to generation of additional charge carriers

Can be due to primary particles or secondary (bremsstrahlung)

 $\sigma = \sigma_0 + k_p D^{\Delta}$

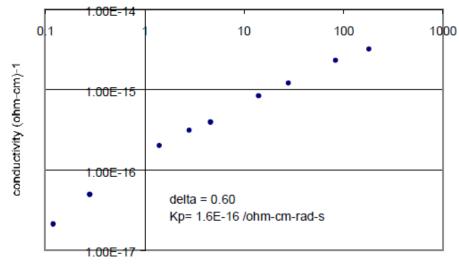
Where σ_0 is the dark conductivity (Ω^{-1} cm⁻¹)

 k_p is the co-efficient of prompt RIC (Ω^{-1} cm⁻¹ rad⁻¹ s)

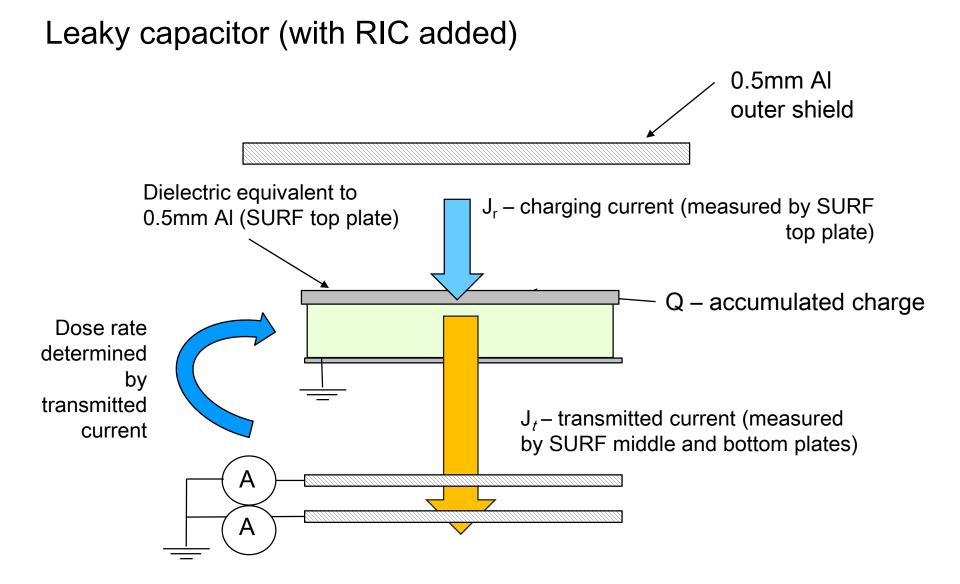
 \dot{D} is ionising dose rate (rad s⁻¹)

 Δ is a dimensionless material dependent exponent (Δ <1)

 Δ is typically in range 0.6 to 1.0



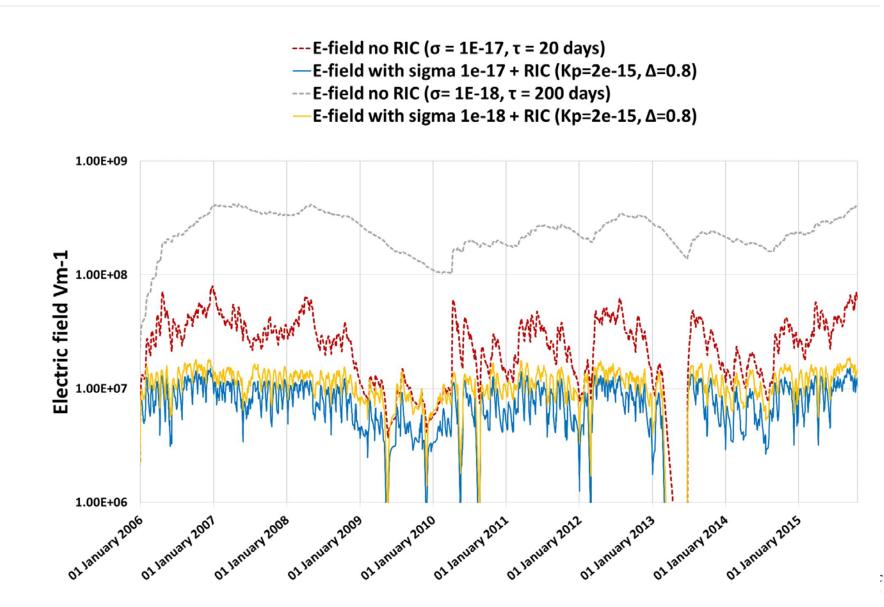
Dose rate (rad/s)



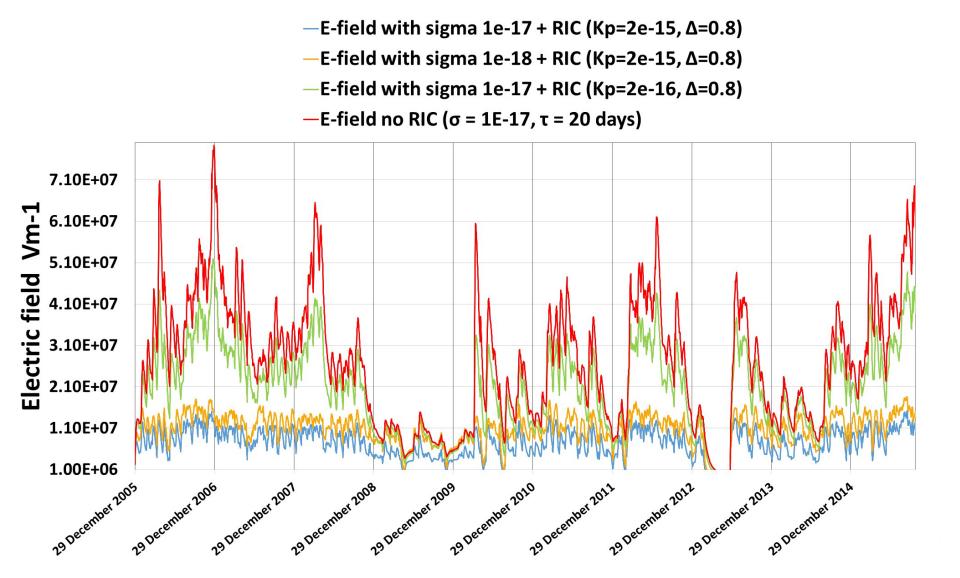
dD/dt [rad/s] = $1.92 \times 10^{11} J_t [A m^{-2}]$



Effect of including RIC



Effect of changing conductivity parameters



Conclusions

Modelling of dielectric internal fields is assisted by using in-flight current measurements

Can take into account RIC: clearly important moderating factor on time constants and field strengths

Need better knowledge of RIC parameters

Need more in-flight charging current measurements

Can generate a space weather ESD threat 'index' – could be done in real time.

